DEPARTMENT OF TRANSPORT AIR SERVICES OTTAWA, CANADA

A-100 Issue 1 Beech Aircraft 3T, 3N 3NM, 3TM April 1, 1970

AIRCRAFT TYPE APPROVAL

Manufacturer

Beach Aircraft Corporation, Wichita, Kansas, U.S.A.

	MODEL 3T (N	ormal Category) Approved March 16, 1970.			
A CONTRACTOR OF THE PARTY OF TH	Engine	2 Pratt & Whitney R985-AN14B Engines with one 4-1/2N and 9N Damper.			
¥ ⁱ a	Fuel	80/87 Octane Minimum Grade Aviation Gasoline.			
	Engine Limits	Maximum Continuous (Sea Level) 34.5 in. Hg., 2200 rpm (400 H.P.) Straight line manifold pressure variation with altitude to 5,000 feet.			
		Take-off (One Mimute) 36.5 in. Hg., 2300 rpm (450 H.P.)			
	Propeller	2 Hamilton Standard Hydromatic Full Feathering 22 D 30 Hub, 65334-218 Blades.			
	Airspeed Limits (IAS)	Manoeuvering 153 mph (133 kts.) Cruising 205 mph (178 kts.) Never Exceed 257 mph (223 kts.) Flaps Extended 120 mph (104 kts.) Landing Gear Extended 125 mph (109 kts.)			
	C.G. Range	(+109.8) to (+117.7) Laming gear extended. Moment due to retraction of landing gear is +12,000 in./lb.			
	Maximum Weight	8,750 lbs.			
	Minimus Crew	One Pilot (+87).			



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DEL 3T (Conted)

No. of Seats

Maximum of 10 (2 Crew at +87).

Maximum Baggage

Nose Compartment - 600 lbs. at (+28). When auxiliary fuel tanks are installed combined weight of fuel and baggage must not exceed 600 lbs.

Control Surface Movements

Wing Flaps	$\mathtt{U}_{\mathtt{p}}$	0.0	Down	440
Elevator Trim Tab	Up .	18°	Down	12°
Elevator	Up	33°	Down	25°
Aileron Trim Tab	$\mathtt{U}_{\mathbf{p}}$	210	Down	20°
Aileron	$\mathtt{U}_{\mathbf{P}}$	400	Down	23°
Rudder Trim Tab	Right	30°	Left	210
Rudder	Right	21°	Left	21°

Serial Nos. Eligible

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43-34450	43-35490	43-35630	43-35863
43-35455	43-35492	43-35632	43-35869
43-35457	43-35493	43-35634	43-35873
43-35465	43-35495	43 - 356 35	43-35877
43-35466	43-35496	43-35639	43-35885
43-35468	43 - 35535	43-35645	
43-35469	43-35536		43-45718
43-35470		43-35651	44-47213
	43-35540	43-35653	44-47215
43-35476	43-35541	43-35668	44-47216
43-35477	43-35548	43-35671	44-47409
43-35478	43-35549	43-35683	44-47428
43-35479	43-35550	43-35690	44-47433
43-35480	43-35551	43-35691	44-47626
43-35481	43-35552	43-35693	44-47630
43-35482	43-35553	43-35707	44-47633
43-35483	43-35556	43-35741	44-47634
43-35484	43-35557	43-35745	44-47694
43-35485	43-35559	43-35768	44-86939
43-35486	43-35625	43-35824	44-86940
43-35487	43-35627	43-35850	11.00710
43-35488	43-35629	43-35861	•(

MODEL 3T (Cont'd)

Airworthiness Approval Basis

Conversion of aircraft operated by the Canadian Military Forces to Civil Standards in accordance with Bristol Engineering Report #2227, or an equivalent conversion acceptable to the Chief Aeronautical Engineer, Department of Transport.

Approved Equipment

The equipment items listed in F.A.A. Specification A-765 as applicable to D-18S aircraft.

II. MODELS 3N, 3NM AND 3TM (Normal Category) Approved March 16, 1970.

Engine

2 Pratt & Whitney R985-AN14B engines with

one 4-1/2N and 9N Damper.

Fuel

80/87 Octane Minimum Grade Aviation

Gasoline.

Beech

Engine Limits

Maximum Continuous

(Sea Level) 34.5 in. Hg., 2200 rpm (400 H.P.)

Straight line manifold pressure variation with

altitude to 5,000 feet.

Take-off (One Minute)

36.5 in. Hg., 2300 rpm (450 H.P.)

Propeller

2 Hamilton Standard Hydromatic Full Feathering

22 D 30 Hub, 6533A-21S Blades.

Airspeed Limits

(IAS)

Manoeuvering 153 mph (133 kts.)
Cruising 205 mph (178 kts.)
Never Exceed 257 mph (223 kts.)
Flaps Extended 120 mph (104 kts.)
Landing Gear Extended 125 mph (109 kts.)

C.G. Range

(+109.8) to (+117.7) Landing Gear Extended. Moment due to retraction of landing gear is

+12,000 in./lb.

Left 25°

MUDELS 3N, 3NM, AND 3TM (Cont'd)

Maximum Weight

8,750 lbs.

Rudder

Beech

Minimum Crew

One Pilot (+87).

No. of Seats

Maximum of 10 (2 Crew at +87).

Maximum Baggago

Nose Compartment - 600 lbs. at (+28). When auxiliary fuel tanks are installed combined weight of fuel and baggage must not exceed 600 lbs.

Control Surface Movements

Wing Flaps	$\mathtt{U}_{\mathtt{p}}$	0 •	Down	450
Elevator Trim Tabs	Up	18°	Down	13°
Elevator	Ūр	35°	Down	25°
Aileron Trim Tabs	Up	22°	Down	19°
Aileron	ŋp	38.5°	Down	21.
Rudder Trim Tabs	Right	28° to 33°	Left	28° to 33°

Right

Serial Nos. Eligible

CA-1 th	uru Cl	1-100	440	3N
CA-101			-	3TM
CA-102	thru	CA-115	4=	3NM
CA-116	thru	CA-135	- ·	3TM
CA-136	thru	CA-155	-	3NM
CA-156	thru	CA-175		3TM
CA-176	thru	CA-194	•	3NM
CA-195	thru	CA-200	••	3TM
CA-201	thru	CA-261	-	3NM
CA-262	thru	CA-280	-	3NM
CA-281			-	3TM

Airworthiness Approval Basis

Conversion of aircraft operated by the Canadian Military Forces to Civil Standards in accordance with Bristol Engineering Report #2227, Beech Drawing 18-5011, or an equivalent conversion acceptable to the Chief Aeronautical Engineer. Department of Transport.

Approved Equipment

The equipment items listed in F.A.A. Specification A-765 as applicable to D-18S aircraft, and to 3N. 3NM and 3TM aircraft.

SPECIFICATIONS PERTINENT TO ALL MODELS

102 in. forward of center line of main wing Datum

Beech

sper (Placard denoting datum installed on

bottom of fuselage).

Levelling lugs on top of fuselage forward of Levelling Means

cabin.

235 Imp. Gal. (282 U.S. Gal.) Usable Fuel Capacity

(4 tanks in wing, 2 - 63 Imp. Gal. (76 U.S. Gal.)

tanks at +126; 2 - 21 Imp. Gal. (25 U.S. Gal.)

tanks at +155; 1 nose tank, 67 Imp. Gal.

(80.5 U.S. Gal.).

Oil Capacity 11.7 Imp. Gal. (14 U.S. Gal.) Usable at (+93).

Anti-Icer Fluid (2 U.S. Gal.) 2.5 Imp. Gal.

(20 lbs. at +97).

A current weight and balance report including list NOTE 1. of equipment included in certificated empty weight must be in each sircraft at the time of original certification and thereafter.

> The certificated empty weight and corresponding centre of gravity location must include the following unusable fuel and oil:

Fuel: (Two 63 Imp. Gal., and

6 lb. (+122) Two 21 Imp. Gal. Wing Tanks)

4 16. (+ 32) Fuel: (One 67 Imp. Gal. Nose Tank)

Oil: (Including 9.5 lb. feathering

oil in each tank) with

Hamilton Standard

22D30-6533A-21S Propellers 52 lb. (+ 68) NOTE 2.

On airplanes with nose compartment baggage provisions on the innerside of nose door:

Beech

"Maximum Structural Capacity of Nose Compartment - 600 lbs."

NOTE 3.

Models 3N, 3NM and 3TM aircraft were manufactured by Beech Aircraft Corporation as basic D18S Aircraft with equipment installations required for Military roles. The Model 3T aircraft was manufactured by Beech Aircraft Corporation as a basic C18S aircraft and later modified in accordance with R.C.A.F. Requirement RM 1-29 and Beech Aircraft Corporation Model Specification. 139.

NOTE 4.

Current D.O.T. and F.A.A. Airworthiness Directives applicable to Beech D18 Type Aircraft are mandatory for Beach 3N, 3NM, 3TM and 3T Aircraft.

NOTE 5.

For modification purposes the design criteria is CAR 03 effective November 1, 1945; CAR 3 effective November 1, 1949, as amended by 3-14 (Paragraph 3.242).

> (W.M. McLeish), Chief Aeronautical Engineer, Department of Transport.

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